22 March 1967

	22 Watch 170;
5X1A	To:
	Dear Ivan,
5X1A	As you know, has been using full pressure suit helmets that have an oral-nasal oxygen mask installed in order to permit him to fly with the face visor up at all times. For various reasons, of which you are aware, he prefers this system to any other and wishes to retain it as his personal equipment whenever it is necessary to fly with the pressure suit in our aircraft. Since his primary flight activities these days concern the Edwards-Palmdale area, we would like you to maintain all of his equipment for him in the future; we will not retain any of his personal equipment at any other base. If flights are required some place other than the Edwards-Palmdale area, he would take his gear with him from your shop.
	The following data has been primarily compiled by and is 25X1 being passed on to you in the event that any of this information may not have been available.
	There have been six of these helmets made which are designated as GN-ACS-1621 Phase II helmets. Four of these are in use and two are still available for another pilot to use in evaluation or test work.
25X1A	has in his possession two helmets that are listed as GN-ACS-1621 Phase II helmets. This particular helmet was ordered by the customer as an evaluation item to eliminate reflections from the gold plated heated visors presently used on all standard GN-ACS-880 full pressure helmets. To my knowledge, the GN-ACS-1621 Phase II helmet is still an evaluation item, as only two pilots, and one other are the only ones to replace their standard GN-ACS-880 full pressure helmets with the Phase II GN-ACS-1621 mask helmet. The GN-ACS-1621 helmet has advantages over the standard helmet because the visor remains in the up position, eliminating reflections from the gold plated visor. Also, the two pilots often remarked about the added cooling from the suit vent air passing up through the suit neck ring past their face.
	The GN-ACS-1621 helmets are fitted with a counter balance spring

The GN-ACS-1621 helmets are fitted with a counter balance spring assembly (GN-A2840) which adds to the closing of the visor under "G" forces, or if the pilot had to leave the aircraft in an inverted position.

All ACS-1621 helmets have aneroid operated visors and are scheduled

Approved For Belease 2002/06/24: CIA-RDP75B00285R000400130005-9 22 March 1967

to close between 29,000 and 31,000 feet which is at least 3,000 feet before the aneroid in the suit controller functions.

25X1A

25X1A

before the anerold in the suit controller functions.	
The ACS-1621 Phase II helmet can be used with either an S970 (rear entry) suit assembly, which has two, or an S901 "U" entry suit assembly.	25X1A
One of Standard with the other existing five helmets of this type. This particular helmet has the original plumbing which connects the two internal oxygen hoses into the helmet manifold. This type of connector is very comfortable as it does not have a 3-1/2 inch long rigid hose connection in the neck ring area. This rigid portion of the hose interferes with head mobility and causes some discomfort in flight. The other five helmets were changed to put the hose connection in the neck area because of a disconnect that occurred at one time during suitingup. This is readily avoided by making the proper hose to manifold connection as must be accomplished in several other areas during suit donning. We would like to have other helme converted back to this internal manifold connection so that all of his equipment is the same. We further believe that the other four helmets should be made the same since pilot comfort is directly involved and since there is no reliability problem.	' 25X1A
Number 265, are the two helmet pressure taps ACS-C1311, used to monitor suit and face or mask cavity differential pressure as the Subject is being tested before a flight. This item was proven to be unsafe for flight, since it was found that due to the high profile of the ACS-C1311 pressure tap, it could be struck by a piece of parachute hardware and spun from the helmet. At a later date, the David Clark personnel static tested the entire assembly and found that the suit would not remain pressurized with a pressure tap missing. We were led to believe that if ejection occurred at a high altitude, the pilot would be in serious trouble. We were then directed to remove all ACS-C1311 pressure taps from all full pressure helmets and replace them with flush mounted GN-P-3399 type pressure taps.	25X1A
There is a disadvantage with the flush mounted GN-P-3399 type pressure taps in that they must be removed to make the pressure helmet check and then reinstalled to seal up the tap. This is done in the cockpit prior to flight and the plugs being very small, could be dropped and thus fall below the seat in the cockpit. David Clark should be advised to make a new pressure tap device for the helmets, such that nothing is removable that could be dropped in the cockpit, but is also flush so that it can not be damaged by being struck by any external	•

Approved For Release 2002/06/24 : CIA-RDP75B00285R000400130005-9

and we would like these to remain installed on

object and thus permit the helmet to lose pressure. In the meantime, the only pressure taps that should be used are the GN-P-3399 type

helmets.25X1A

22 March 1967

When the ACS-1621 Phase II helmet is used with either an S970 or an S901 suit assembly, a neck dam GN-A2797 must be installed in the neck ring area of the suit. The neck dam normally remains loose around the Subject's neck to allow adequate vent and comfort for the Subject. The neck dam can be drawn tight by means of a draw string that attaches to the pilot's mask with a snap. In event the Subject had to eject over water, he should draw his neck dam tight before contact with the water as part of his over water ejection procedures. This would prevent water entering the suit through the helmet.

25X1A		You will note another non-standard modification in suspension system. You are	25X1A
20/1/		probably well aware of this type of modification. We like it very much	
į		from a comfort and communications standpoint and highly recommend it for use in all of these full pressure suit helmets.	
		·	
		To pre-flight or test this helmet for any reason, you will follow the same procedures as testing an 880 type helmet. There is one change	
		in the test plate used in the mask. The nomenclature and number for	
		this item is GN-IJ-D1125 plate, testing, mask.	
:		NOTE: When testing the GN-ACS-1621 Phase II helmet either in pre-	
		flight, subject test or post-flight, always test the helmet with the visor in the closed position so the complete oxygen system is functioning	
		while the test is being performed.	
		One other test that should be made at least every thirty days is the	
		function of the aneroid operated visor in a chamber at altitude. The	
		aneroid should allow the visor to close between 29,000 and 31,000 feet.	
		After the test, the aneroid can easily be re-armed manually.	
		A number of spare parts used only on Phase II helmets should be kept	•
		in stock and installed only by experienced personnel. Enclosed are thirty day periodic and annual IRAN forms that we presently use, which	
05V4A		cover the testing and inspection of all David Clark pilot's protective	25X1A
25X1A		assemblies, including	
•			
•	,	<u> </u>	
	•	Best regards,	25X1A
			7
	· · · · · · · · · · · · · · · · · · ·		
			<u>}</u>
25X1A		·	
		•	

Approved For Release 2002/06/24: CIA-RDF75B00285R000400130005-9

Approved For Release 2002/06/24 : CIA-RDP75B00285R000400130005-9

ANNUAL TRAN SUIT ASSEMBLY

PAGE:

ASSEMBLY NO.	DATE INSPECTED		
	SIGNATURE		
HELMET ASSEMBLY			
A. OXYGEN SYSTEM PLUMBING STRESS TEST	@125 PSI FOR 2 MINUTES		
B. OXYGEN SYSTEM LEAK RATE	@70 PSI ******** LEAK		
C. HELMET SHELL LEAK RATE	@3.5 PSI 700 cc/m MAX.		
D. FACE BARRIER LEAK RATE WITH EXHALATION VALVE	400 cc/m MAXIMUM LEAK RATE		
E. PNUEMATIC SEAL BLEED DOWN TIME. NO 1 NO2.	BOTH 45-60 SECONDS		
F. LENS RESISTANCE TEST, UNPRESSURIZED PRESS	URIZED 20.3 +- 3.6 OHMS		
G. COMMUNICATION TEST			
E. ON PHASE 11 HELMETS HAVE ANEROID FIRED IN CHAMBER	AND RECORD MAX. 3),000		
BASIC SUIT TEST.			
A. BASIC SUIT STRESS TEST, WITH HELMET AND GLOVES TI	LL RELIEF VALVE ACCTUATES		
B. RELIEF VALVE TEST, LEFT RIGHT	@4 TO 5 PSI		
C. BASIC SUIT LEAK RATE	@ 3.5 PSI 850 cc/m MAX.		
FLOATATION GARMENT TEST.			
A. INFLATE GARMENT TO 12" H20 FOR 10 MINUTES DROP TO) 11" H2O		
B. LEAVE INFLATED FOR 1 HOUR. MUST NOT LOOSE OVER 3"	H2O IN THAT HOUR		
SUIT ASSEMBLY PRESS TO TEST	•		
A. SYSTEM 1. MAXIMUM PRESSURE MMHG	110 +- 10 MMHG		
. DIFFERENTIAL INCHES	1.0 TO 1.5 H2O		
LRAK RATE CC/M	1500 cc MAXIMUM		
B. SYSTEM 2. MAXIMUM PRESSURE MMHG	•		
DIFFERENTIAL INCHES			
LBAK RATE CC/M	1500 cc MAXIMUM		
C. COMPLETE OXYGEN SYSTEM LEAK RATE UNPRESSURIZED_	600 cc MAXIMUM		

ANNUAL IRAN INSPECTION

HELMET				
A. INSPECT CLOSING OF BOTH SUNSHADE AND CLEAR VISORS AND OPERATION OF LEFT HAN	D BEARING			
B. CLEAN BOTH VISORS AND CHECK AND REMOVE SCRATCHES IF APPLICABLE				
C. INSPECT FACE HEAT WIRING				
D. INSPECT COMMUNICATION WIRING FOR SOLID CONNECTIONS BOTH SOLDER AND SCREW TY	PE			
· -				
E. INSPECT FACE SEAL REPLACE IF NECESSARY				
F. INSPECT HELMET RACE , AND WIPER SEAL, CLEAN OR REPLACE				
G. INSPECT HELEMT HOSES FOR WEAR, HAVE REPLACED IF NEEDED				
H. INSPECT WIND-UP KNOB FOR LOOSE SCREWS AND OPERATION OF TAKE UP				
RESTRAINT				
A. INSPECT GARMENT FOR WEAR, TEARS, AND SMOOTHNESS OF ALL ZIPPERS				
B. INSPECT ALL SCREWS FOR BURRS AND TIGHTNESS (TORQUE ALL SCREWS PER SPEC.				
C. INSPECT SUIT HALF NECK RING DONNING CLIP, LOCKING LATCH, STATIC SEAL				
D. INSPECT PRESSURE SEALING ZIPPERS, CLEAN AND LUBRICATE				
E. INSPECT SUIT HALF GLOVE DISCONNECTS. REPLACE LEFT AND RIGHT "O" RINGS	,			
F. INSPECT HOLD-DOWN ASSEMBLY. INSPECT HOLD-DOWN LANYARD FOR WEAR				
GLOVES,				
A. INSPECT HARDWEAR SCREWS FOR TORQUE. REMOVE AND REPLACE SCREWS WITH BURRS				
B. INSPECT STITCHING, PALM RESTRAINTS, TAPING AND ZIPPERS				
C. INSPECT CURE DATE OF GLOVE BLADDERS AND REPLACE IF DUE				
OUTER GARMENT.				
A. INSPECT ALL ZIPPERS, FLAPS, POCKETS AND STITCHING				
B. INSPECT ALL SNAPS FOR TIGHTNESS. REPLACE IF NEEDED				
C. INSPECT FLOATATION GARMENT LACING, WEIGHT OF CO2 BOTTLE, RELACE ALL "O"				
RINGS AND RESAFETY ACCUATOR WITH "E" THREAD. INSERT CO2 BOTTLE AND TORQUE				
WITH 60 LBS. LOCK IN PLACE WITH SAFETY SCREW.				

	PAGE 3	Approved For Release 2002/06/24 : CIA-RDP75B00285R000400130005-9		
	VENT INPUT	ASSEMBLY	September (1985) de l'Alle Carrelle de l'Alle de l L'Alle de l'Alle de l'All	_
	A. REPLACE	FLAPPER VALVE AND "O" RING IN VENT INPUT ASSEMBLY		
	B. INSPECT	ALL SCREWS FOR BURRS. REPLACE IF NEEDED AND TORQUE ALL SCREWS.		
	SUIT ALTIM	ETER (IF APPLICABLE)		-
	A. INSPECT	ALTIMETER LENS FOR SCRATCHES OR CRACKS. REPLACE LENS IF NEEDED	-	
	B. INSPECT	AND TORQUE ALL SCREWS ON MOUNTING PLATE.		
	BOOTS			_
	A. INSPEC	T ZIPPERS, LACING AND PERMANENT ATTACHED SPURS IF APPLICABLE.	***	
-	B. POLISH	BOOTS.	-	
	REMARKS			
	HELMET			
	RESTRAINT			
	GLOVES			
	OUTER GAR	MENT		
	BOOTS			
	SIGNATURE	OF INSPECTING TECHNICIAN_		

Approved For Release 2002/06/24: CIA-RDP75B00285R000400130005-9 5970 PERIODIC INSPECTION S901 (E) DATE SUIT ASSY TECH TYPE INSPECTION: 30 days elapsed time______ 5 flights____ 1. Stress O2 system to 125 PSI for 1 minute 2, 0, system leak rate (garment unpressurized) Input pressure 70 PSI CC/M 800 CC/M Max BOTH SYSTEMS . . 3. Press-to-test Check 'MM System #1 MAX Press 110 + 10 MMHg. "H₂0 Differential 1.0 - 1.5" H20 CCM Leak Rate 1800 CC/M Max MM '' System #2 ~"H_0 CCM 4. Stress suit assy to point at which relief valves "POP" PSI 1 Min duration (4.5 ± 0.5 PSI) MAX PSI MIN CC/M 5. Leak Rate @ 3.5 PSI - 182 MMHg MAX -2000 CC/M 6. Lens Resistance, w/pressure, wo/pressure OHMS 7. Anti-suffocation Bleed Rate 30-60 mm seconds . . . BOTH SYSTEMS _ _____SEC 8. Communications Check 9. Visual Inspection PERIODIC VISUAL INSPECTION CHECK (Page 2) COMMENTS:

Approved For Release 2002/06/24 CIA-RDP75B00285R000400130005-9

PERIODIC VISUAL INSPECTION

HELMIST	л.	oneck closing of both visors and operation of latening knobs.	
	В.	Clean both visors and check for scratches.	
•	C.	Clean heater contacts.	
	D.	Check wiring and communications plug.	
	E.	Check face seal-or MASK	
	F.	Check neck bearing torque.	
RESTRAINT	D	Check hoses for kinks, tightness, etc. Check Helmet angroid of Applicable Inspect garment for wear, rips, tears, etc.	
	В.	Inspect all zippers.	
	c.	Check lacing of restraint cover.	
	D.	Check all screws for tightness.	
	E.	Check helmet disconnect; Back clip, O-ring, ring latch.	
,	F.	Forearm disconnect 0-ring.	
. •	G. :	Helmet hold-down adjustment strap, cable, screws for tightness, Tru-Arc locks.	
GLOVES	A.	Check screws on flanges for tightness.	
	В.	Check stitching.	F- 251
	C.	Check palm restraints.	12. 725. /
	D.	Check zippers.	
	E.	Check taping on inside of disconnect.	
BOOTS	A.	Check zippers.	
	в.	Check lacing.	
	C.	Polish boots if needed.	
COVERALL	A.	Check all zippers and flaps.	
	В.	Check all snaps.	
	c.	Check flotation gear lacing.	
	D,	Weight flotation bottle.	
	E.	Check safety seal.	
	F.	Check Floatation Garment	